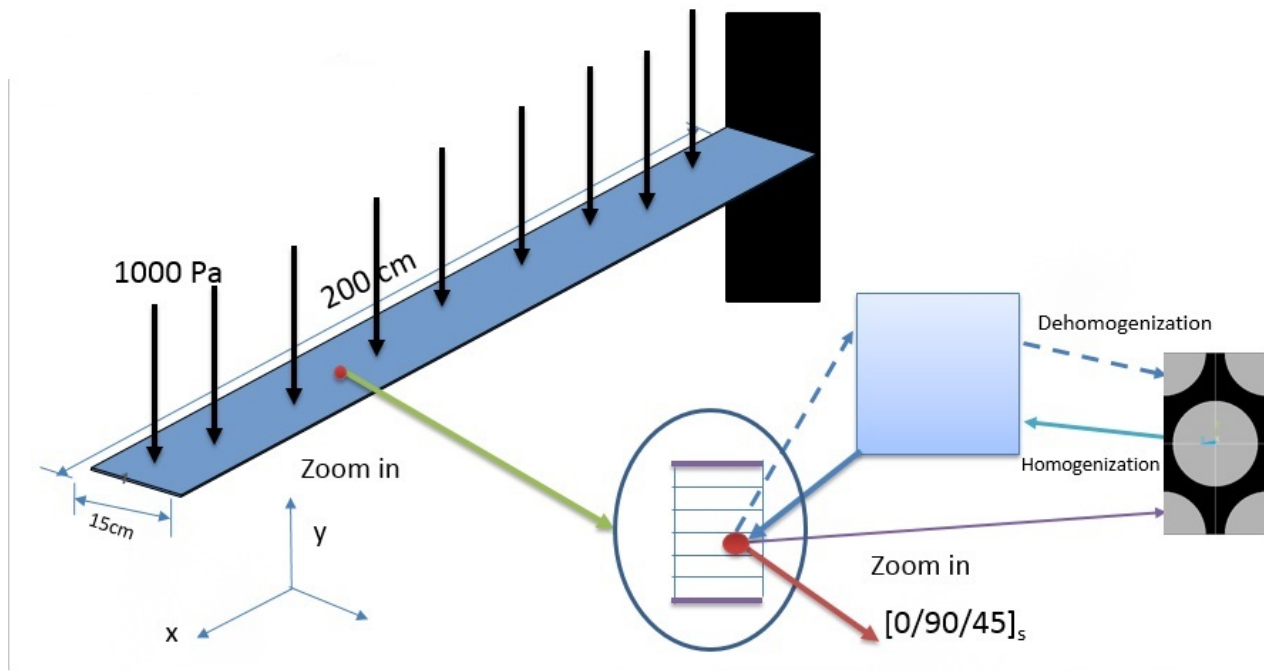


## Predictions of local/global stress/strain fields in composite structures

In this problem, we will try to show how to analyze the local-global fields in composite structures using [SwiftComp](#)-Abaqus-GUI.

The figure below can summarize how to do the local global analysis.



Let the material properties a fiber (T300) property be: :  $E_{11}=230$  GPA,  $E_{22}=15$  GPA,  $\nu_{12}=0.20$ ,  $\nu_{23}=0.0714$ ,  $G_{12}=15$ GPa,  $G_7=3.928$ GPa.

and matrix (3501-6 epoxy) be:  $E =4.2$ GP,  $\nu=0.34$

The composite lay-up:

$[0/90/45]_{, , s , ,}$

Thickness of each ply=0.00025m


“Soden, P. D., Hinton M. J. and Kaddour, A. S., Lamina properties, lay-up configurations and loading conditions for a range of fibre reinforced composite laminates. Compos. Sci. Technol., 1998, 58(7), 1011”

Major steps to perform local-global analysis

### Step 1: Input material properties

There are two materials namely fiber and matrix

- a. Fiber properties
- b. Matrix properties

 Edit Material
✕

Name: Fiber

Description:

Material Behaviors

Density

Elastic

General
Mechanical
Thermal
Electrical/Magnetic
Other

Elastic

Type: Engineering Constants ▼ Suboptions

Use temperature-dependent data

Number of field variables: 0

Moduli time scale (for viscoelasticity): Long-term

No compression

No tension

Data

	E1	E2	E3	Nu12	Nu13
1	230000000000	15000000000	15000000000	0.2	0.2

OK
Cancel

✚ Edit Material
✕

Name: Matrix

Description:

Material Behaviors

Density

Elastic

General
Mechanical
Thermal
Electrical/Magnetic
Other

Elastic

Type: Isotropic ▼ Suboptions

Use temperature-dependent data

Number of field variables: 0

Moduli time scale (for viscoelasticity): Long-term

No compression

No tension

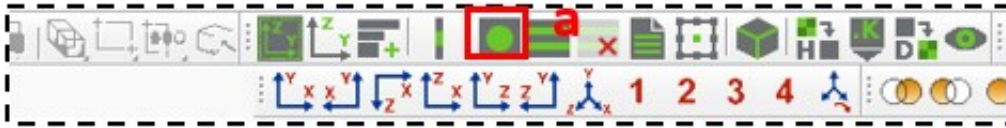
Data

	Young's Modulus	Poisson's Ratio
1	4200000000	0.34

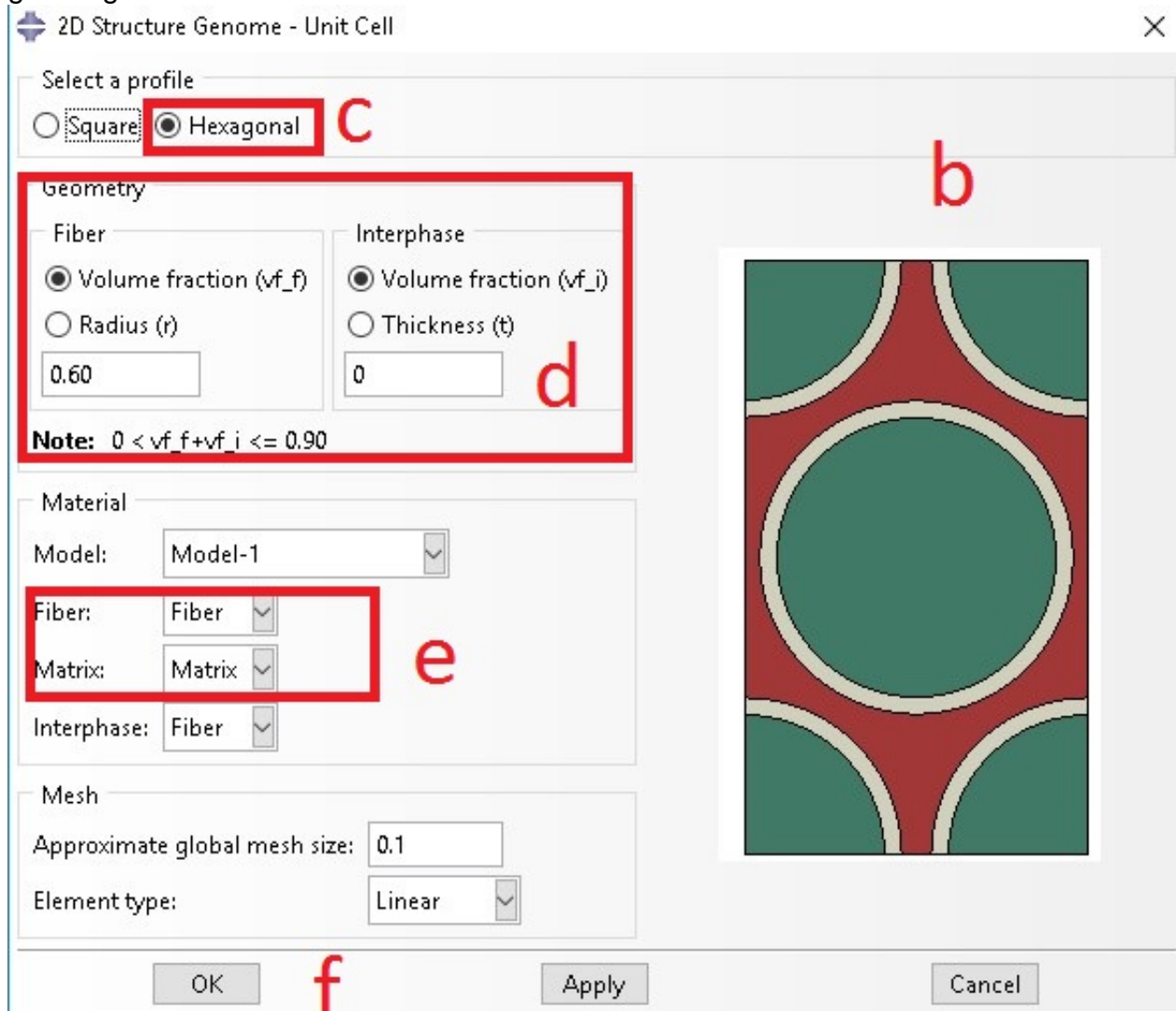
OK
Cancel

**Step 2: Select appropriate SG**

- a. Select 3D SG that represent the current example



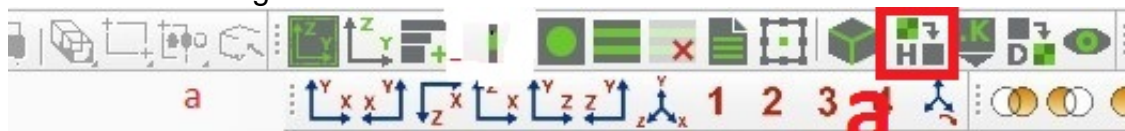
- b. 3D SG wizard shows up
- c. Select spherical inclusion as microstructure
- d. Add inclusion volume fraction
- e. Select material properties for inclusion and matrix
- f. Click on OK to generate the SG
- g. See generated 2D SG



(Image(Problem-4bb.JPG) failed - File not found)

**Step 3- Homogenization- 3D effective properties**

- a. Click on Homogenization



- b. Homogenization wizard shows up ( see below)
- c. Select 3D (solid) Model
- d. Select analysis type, elastic
- e. Click on OK to start homogenization
- f. See the predicted 3D effective properties

➔ Homogenization

New SwiftComp file name:

Model source

CAE  Input file

Model:  Part:

Macroscopic model

Dimension

1D (Beam)

2D (Shell)

3D (Solid)

Dimensionally reducible structures

Specific model:

Omega:

**Note:** Provide omega if the part is not a line, rectangle or cube

Options

Analysis type:

Element type:

Elemental orientation:

Temperature distribution:

Aperiodic

y1  y2  y3

Only generate input file. Do not run SwiftComp.

b

c

d

e

**Step 4: Export predicted effective properties to create a new model**

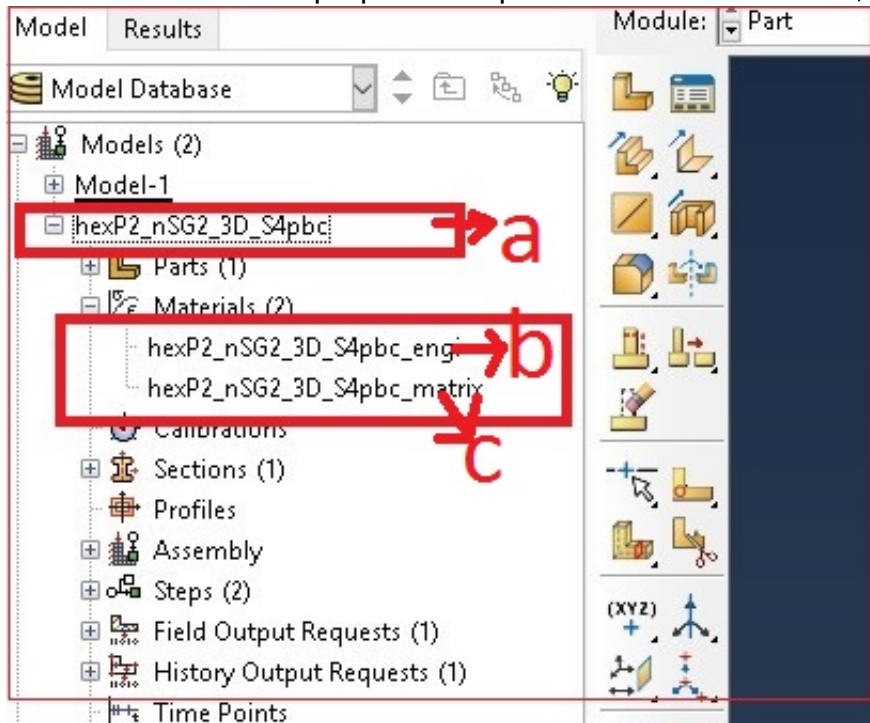
a. Export the predicted effective properties



b. A new model is automatically generated, this is to be used for generating a global model

c. Predicted effective properties exported as engineering constants

d. Predicted effective properties exported as stiffness matrix, D,

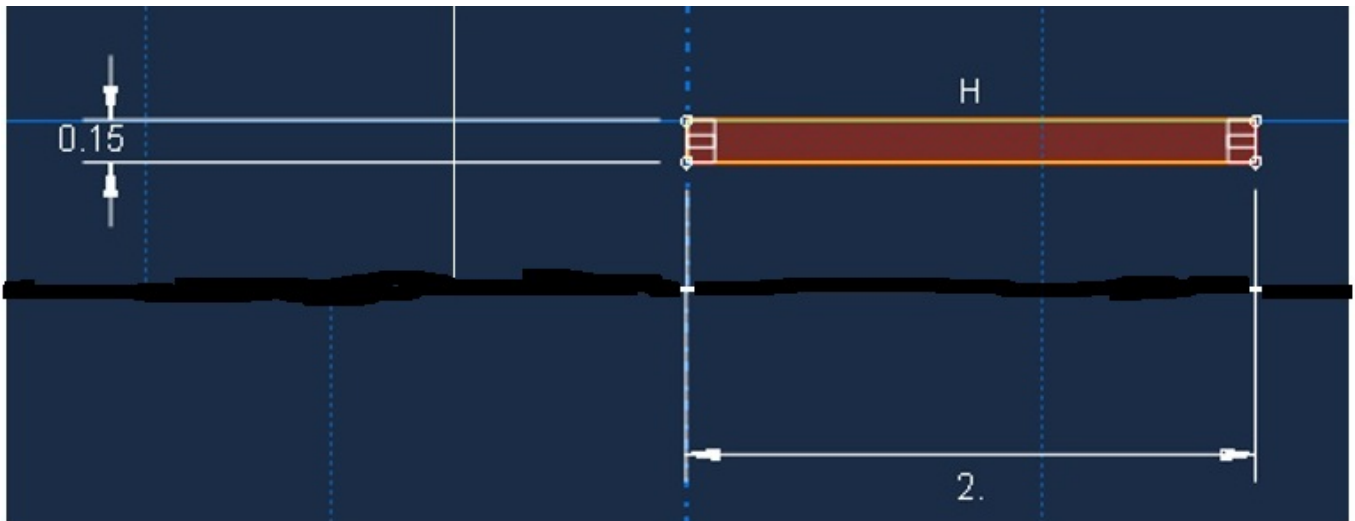
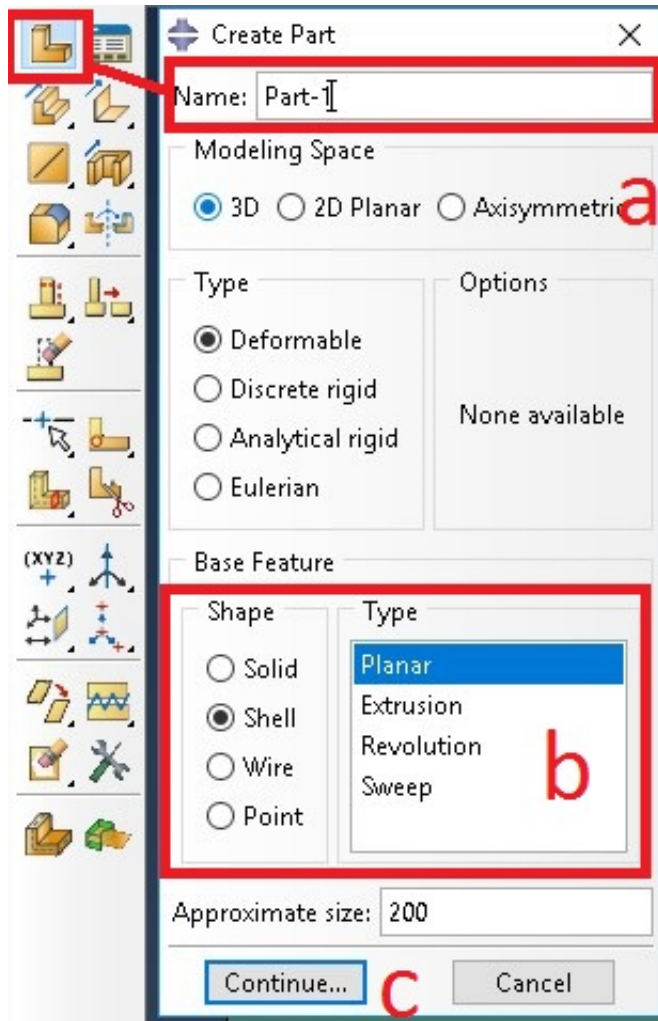


**Step 5: Generate the global model**

a. Click on Part and name as Part-1

b. Select 'Shell' from from Shape and 'Planer' from type

c. A new model is generated as shown ( 0.2m and thickness=0.15m)



*Step 6: Create and Assign Section'*

- Click on Section and name as 'Composite'
- Select 'Shell' from from Category and 'Composite' from type
- Edit section wizard shows up
- Add section properties as shown and click OK

## PREDICTIONS OF LOCAL/GLOBAL STRESS/STRAIN FIELDS IN COMPOSITE STRUCTURES

Create Section

Name: Composite **a**

Category:  Solid  Shell  Beam  Fluid  Other

Type: Homogeneous Composite **b** Membrane Surface General Shell Stiffness

Continue... Cancel

Edit Section

Name: CompositeA

Type: Shell / Continuum Shell, Composite

Section integration:  During analysis  Before analysis

Layup name:

Basic Advanced

Thickness integration rule:  Simpson  Gauss

Symmetric layers

Material	Thickness	Orientation Angle	Integration Points	Ply Name
hexP2_nSG2_3D_S4pbc_engi	0.00025	0	3	A
hexP2_nSG2_3D_S4pbc_engi	0.00025	90	3	B
hexP2_nSG2_3D_S4pbc_engi	0.00025	45	3	C
hexP2_nSG2_3D_S4pbc_engi	0.00025	45	3	D
hexP2_nSG2_3D_S4pbc_engi	0.00025	90	3	E
hexP2_nSG2_3D_S4pbc_engi	0.00025	0	3	F

Options:

OK Cancel

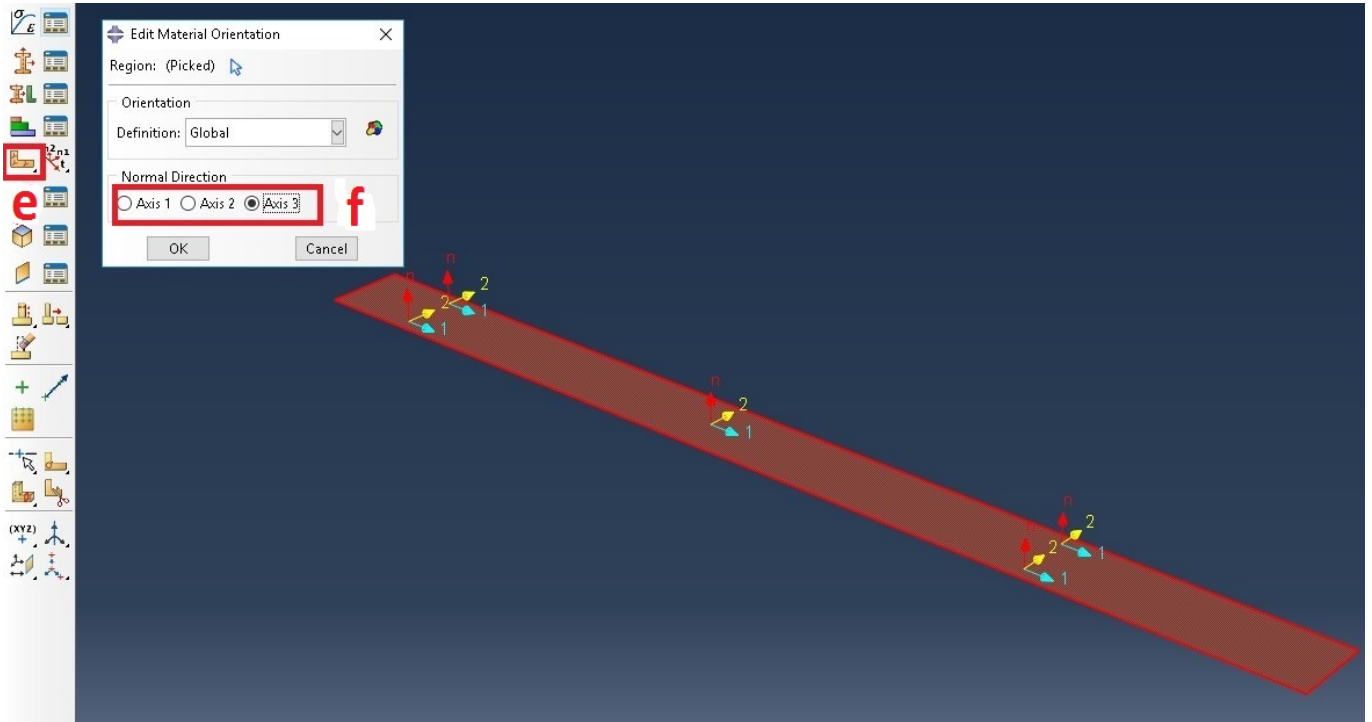
**d**

**e**

- e. Assign material orientation
- f. Click on Axis 3



# PREDICTIONS OF LOCAL/GLOBAL STRESS/STRAIN FIELDS IN COMPOSITE STRUCTURES

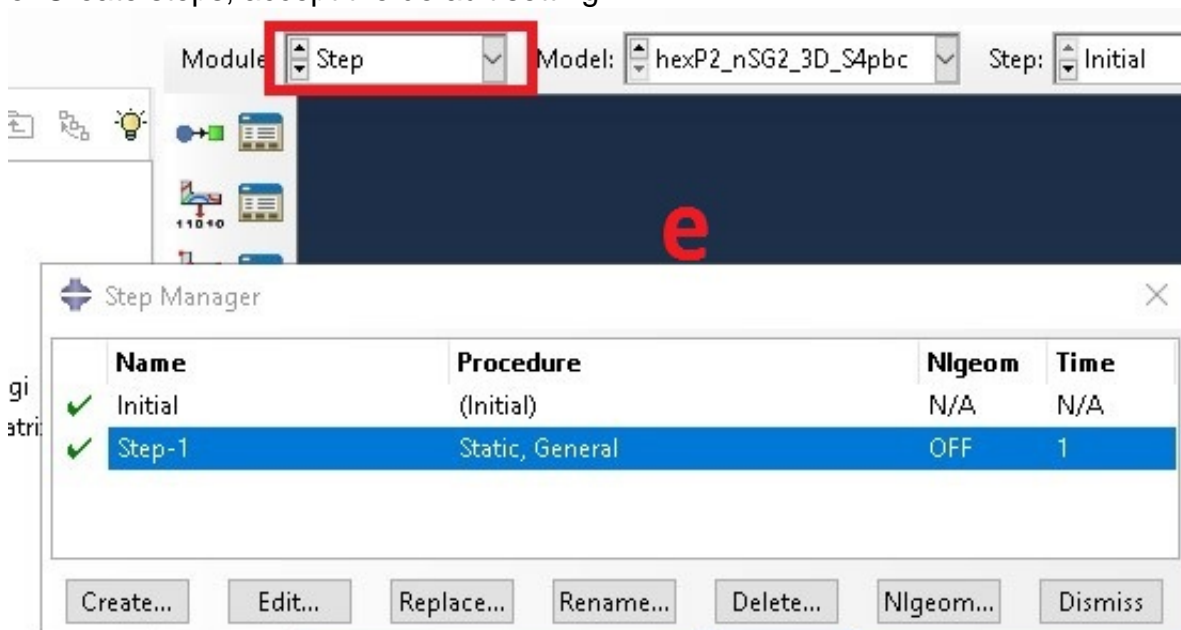


## Step 7: Create Assembly and Steps ‘

- Select Assembly
- Select ‘Part-1’ from Create Instance
- Click OK

(Image(Problem-4M7.jpg) failed - File not found)

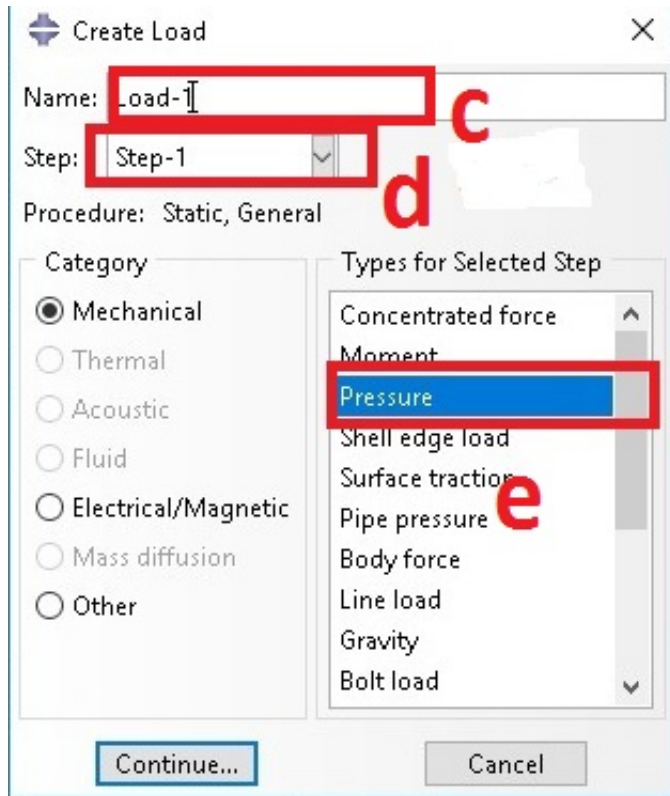
- Create steps, accept the default setting



## Step 8: Create load and boundary conditions ‘

- Select load from Module
- Select load
- Name the load as ‘Load-1’

- d. Select step 1
- e. Select pressure

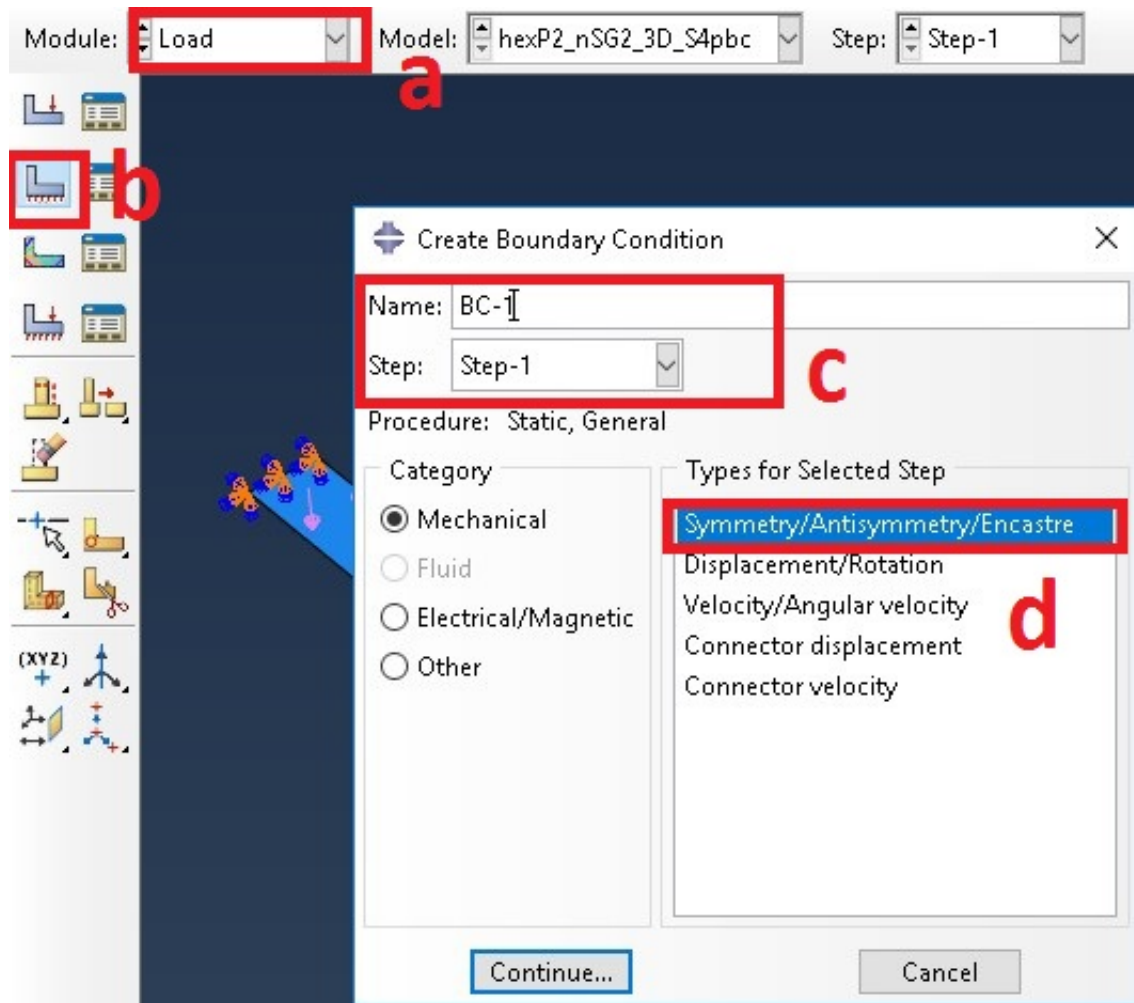


- f. Select the area to be loaded
- g. Add the load

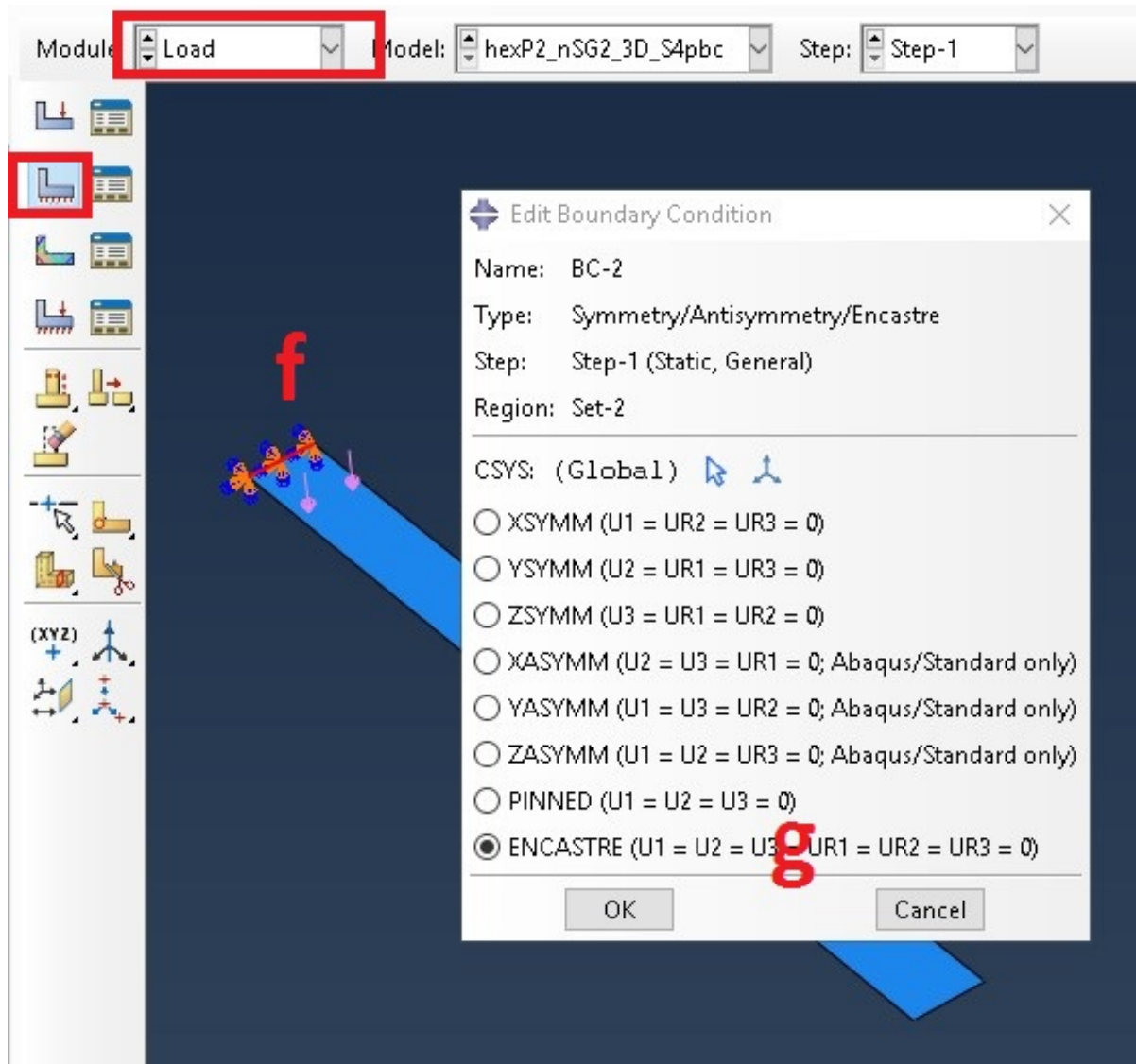
youtube link:

*Step 9: Create boundary conditions*

- a. Select load from Module
- b. Boundary Conditions (BC)
- c. Name the BC as 'BC-1' and Select 'Step-1'
- d. Select 'Mechanical' from Category and Symmetry type BC
- e. Click on OK



- f. Select the edge for BC
- g. Select ENCASTER

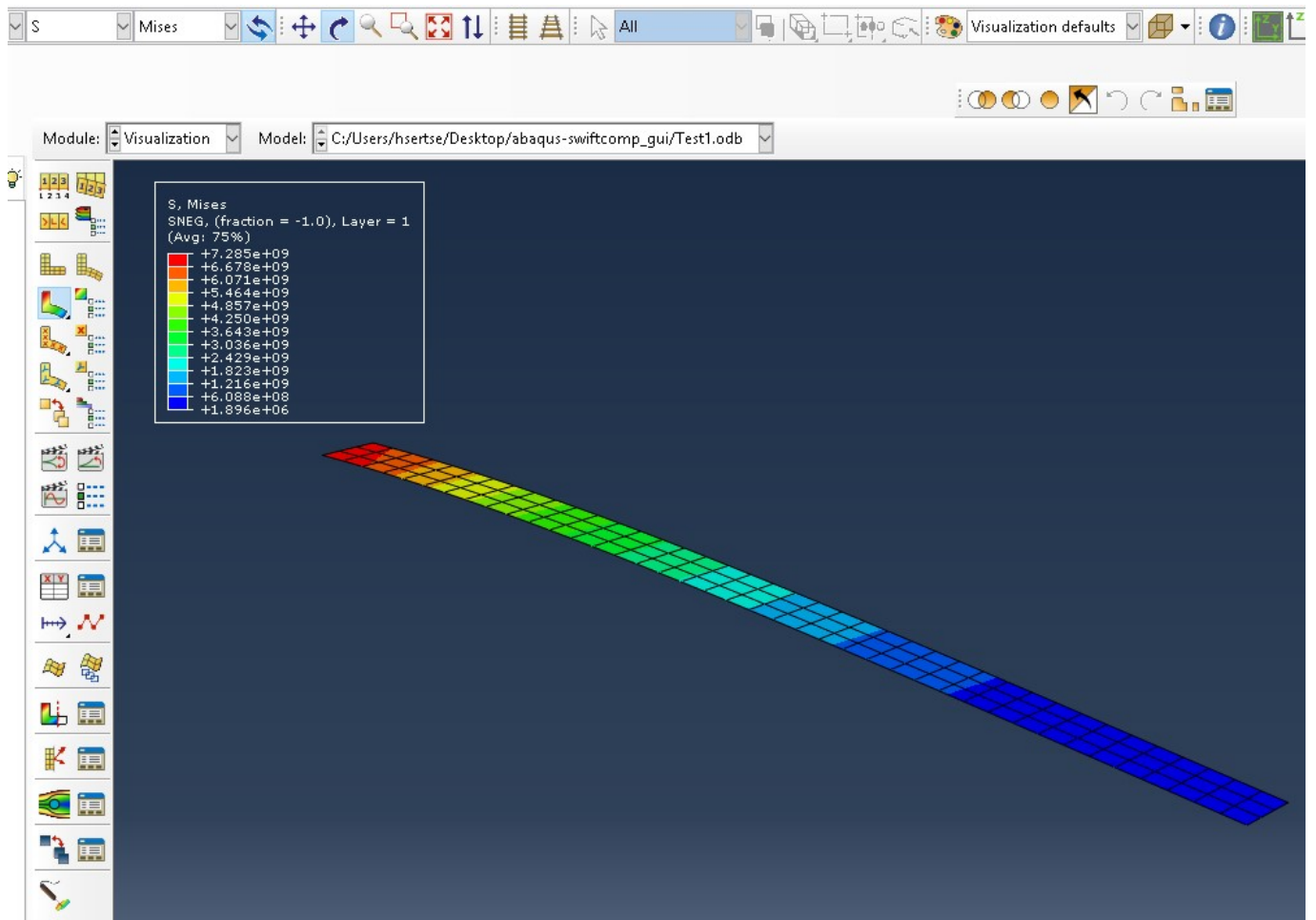


Step 10: Create Mesh ‘

Step 11: Run the analysis ‘

Step 11: Results of the analysis ‘

# PREDICTIONS OF LOCAL/GLOBAL STRESS/STRAIN FIELDS IN COMPOSITE STRUCTURES

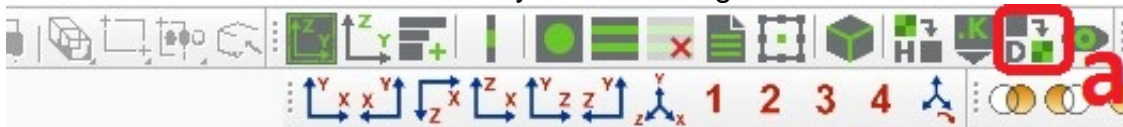


## Step 12: Obtain global strains ‘

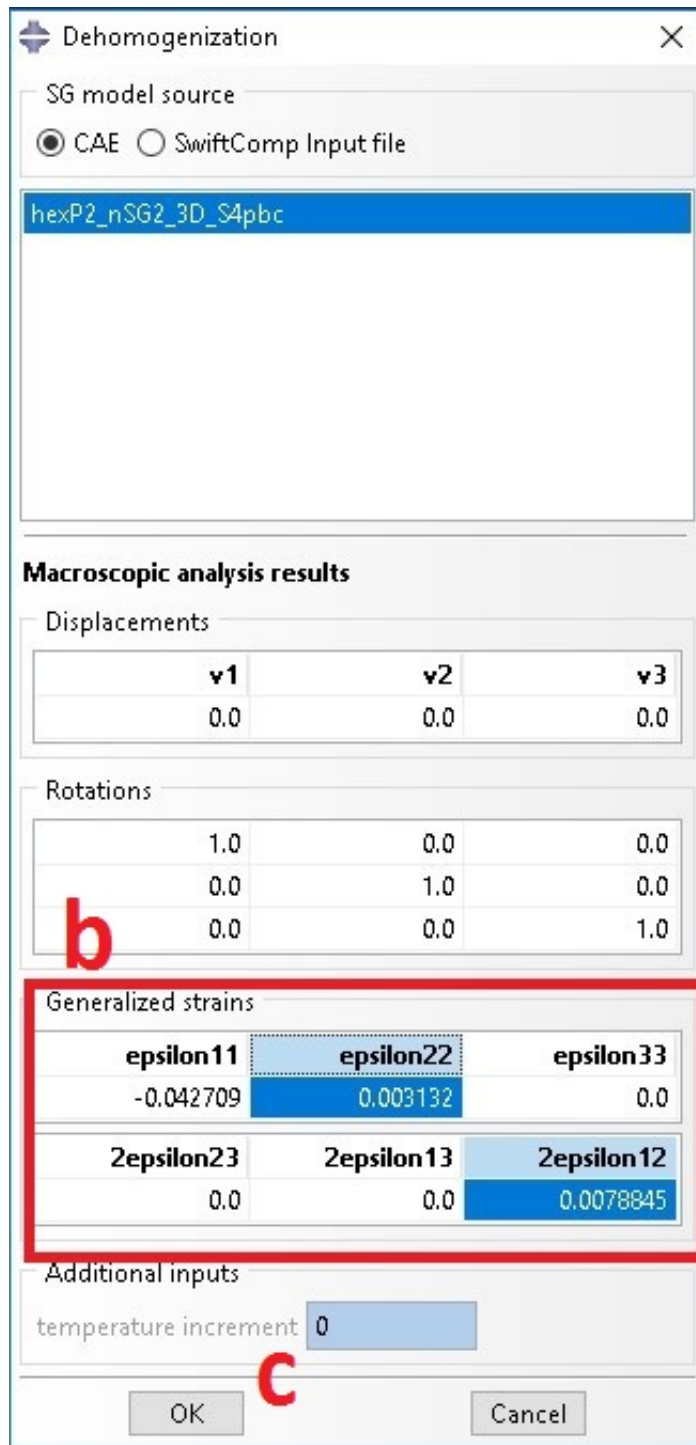
- Click on probe values to obtain global strain
- Click on a point on global structure
- Select nodes and all direct from probe values wizard
- The following strain values can be obtained, strain  $e_{11}=-0.042709$ ,  $e_{22}=0.00212116$ ,  $2e_{12}=0.00788458$ , and all others are zero

## Step 13: Run dehomogenization ‘

- Go back to micromechanical analysis with hexagonal SG and click on dehomogenization



- Add global strain obtain in step 12 to obtain local field in 0 degree lamina
- Click on OK



Step 14: Create view port and show both global and micromechanical local field analysis ‘

# PREDICTIONS OF LOCAL/GLOBAL STRESS/STRAIN FIELDS IN COMPOSITE STRUCTURES

